Program: RFEM 5
Category: Geometrically Linear Analysis, Isotropic Nonlinear Elasticity, Isotropic Plasticity, Member, Plate

## Verification Example: 0018 - Plastic Bending - Tapered Cantilever

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## Description

A tapered cantilever is fully fixed on the left end $(x=0)$ and subjected to a continuous load $q$ according to the Figure 1. Small deformations are considered and the self-weight is neglected in this example. Determine the maximum deflection $u_{z, \max }$. The problem is described by the following set of parameters.

| Material | Elastic-Plastic | Modulus of Elasticity | E | 210000.000 | MPa |
| :---: | :---: | :---: | :---: | :---: | :---: |
|  |  | Poisson's <br> Ratio | $\nu$ | 0.000 | - |
|  |  | Shear <br> Modulus | G | 105000.000 | MPa |
|  |  | Plastic <br> Strength | $f_{y}$ | 240.000 | MPa |
| Geometry | Cantilever | Length | L | 4.000 | m |
|  |  | Width | w | 0.005 | m |
|  |  | Left Side Height | $h_{1}$ | 0.250 | m |
|  |  | Right Side <br> Height | $h_{2}$ | 0.150 | m |
| Load |  | Continuous <br> Load | $q$ | 2300.000 | $\mathrm{Nm}^{-1}$ |



Figure 1: Problem sketch

## Analytical Solution

This is more complex variant of the verification example 0006. The tapered cantilever is considered in this case. The function of the cantilever height $h(x)$ is following

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$$
\begin{equation*}
h(x)=h_{1}+\frac{x}{L}\left(h_{2}-h_{1}\right) \tag{18-1}
\end{equation*}
$$

The bending moment $M$ for the plate under continuous loading $q$ is defined as

$$
\begin{equation*}
M=-\frac{q(L-x)^{2}}{2} \tag{18-2}
\end{equation*}
$$

## Linear Analysis

Considering linear analysis (only elasticity) the maximum deflection of the structure can be calculated as follows:

$$
\begin{equation*}
u_{z, \max }=\int_{0}^{L} \frac{q(L-x)^{3}}{2 E I_{y}(x)} \mathrm{d} x=71.614 \mathrm{~mm} \tag{18-3}
\end{equation*}
$$

## Nonlinear Analysis

The quantities of the load are discussed at first. The maximum bending moment obviously occurs on the fully fixed end. The moment $M_{\mathrm{e}}$ when the first yield occurs and the ultimate moment $M_{\mathrm{p}}$ when the structure becomes plastic hinge are calculated as follows

$$
\begin{align*}
& M_{\mathrm{e}}=2 \int_{0}^{h_{1} / 2} \sigma(z) z w \mathrm{~d} z=2 \int_{0}^{h_{1} / 2} \frac{2 f_{\mathrm{y}}}{t} z^{2} w \mathrm{~d} z=\frac{f_{\mathrm{y}} w h_{1}^{2}}{6}=12.500 \mathrm{kNm}  \tag{18-4}\\
& M_{\mathrm{p}}=2 \int_{0}^{h_{1} / 2} \sigma(z) z w \mathrm{~d} z=2 \int_{0}^{h_{1} / 2} f_{\mathrm{y}} z w \mathrm{~d} z=\frac{f_{\mathrm{y}} w h_{1}^{2}}{4}=18.750 \mathrm{kNm} \tag{18-5}
\end{align*}
$$

The corresponding continuous load $q_{\mathrm{e}}$ and $q_{\mathrm{p}}$ then results

$$
\begin{equation*}
q_{\mathrm{e}}=\frac{2 M_{\mathrm{e}}}{L^{2}}=2.344 \mathrm{Nmm}^{-1} \tag{18-6}
\end{equation*}
$$

$$
\begin{equation*}
q_{\mathrm{p}}=\frac{2 M_{\mathrm{p}}}{L^{2}}=1.563 \mathrm{Nmm}^{-1} \tag{18-7}
\end{equation*}
$$

It is obvious that the continuous load $q$ causes the elastic-plastic state of the plate according to the Figure 1. The bending stress is defined according to the following formula

$$
\begin{equation*}
\sigma_{x}(x, z)=-\kappa(x) E z \tag{18-8}
\end{equation*}
$$

where $\kappa(x)$ is the curvature defined as $\kappa(x)=\mathrm{d}^{2} u_{z} / \mathrm{d} x^{2}$ [1]. The elastic-plastic zone length is described by the parameter $x_{p}$ according to the Figure 1. The bending stress quantity on the

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surface $(z=-h(x) / 2)$ equals to the plastic strength $f_{y}$ at the point $x=x_{p}$, see Figure 2. The curvature at this point can be calculated according to the formula

$$
\begin{equation*}
\kappa\left(x_{\mathrm{p}}\right)=\frac{2 f_{\mathrm{y}}}{\operatorname{Eh}\left(x_{\mathrm{p}}\right)} \tag{18-9}
\end{equation*}
$$



Figure 2: Bending stress distribution
The elastic-plastic moment at the point $x=x_{\mathrm{p}}$ is then

$$
\begin{equation*}
M_{\mathrm{ep}}\left(x_{\mathrm{p}}\right)=\int_{-h\left(x_{\mathrm{p}}\right) / 2}^{h\left(x_{\mathrm{p}}\right) / 2} \sigma_{x}\left(x_{\mathrm{p}}, z\right) z w \mathrm{~d} z=2 \int_{0}^{h\left(x_{\mathrm{p}}\right) / 2}-\frac{2 f_{\mathrm{y}}}{h\left(x_{\mathrm{p}}\right)} z^{2} w \mathrm{~d} z=-\frac{f_{\mathrm{y}} w\left[h\left(x_{\mathrm{p}}\right)\right]^{2}}{6} \tag{18-10}
\end{equation*}
$$

The elastic-plastic moment $M_{\mathrm{ep}}\left(x_{\mathrm{p}}\right)$ (internal force) has to equal to the bending moment $M\left(x_{\mathrm{p}}\right)$ (external force).

$$
\begin{equation*}
-\frac{f_{y} w\left[h\left(x_{p}\right)\right]^{2}}{6}=-\frac{q\left(L-x_{p}\right)^{2}}{2} \tag{18-11}
\end{equation*}
$$

The elastic-plastic zone length $x_{p}$ results from this equality as follows

$$
\begin{equation*}
x_{\mathrm{p}}=\frac{L-h_{1} \sqrt{\frac{f_{\mathrm{y}} w}{3 q}}}{1+\frac{h_{2}-h_{1}}{L} \sqrt{\frac{f_{\mathrm{y}} w}{3 q}}}=1048.915 \mathrm{~mm} \tag{18-12}
\end{equation*}
$$

The curvature $\kappa_{\mathrm{e}}$ in the elastic zone $\left(x>x_{\mathrm{p}}\right)$ is described by the Bernoulli-Euler formula

$$
\begin{equation*}
\kappa_{\mathrm{e}}=-\frac{M}{E I_{y}(x)}=\frac{q(L-x)^{2}}{2 E I_{y}(x)} \tag{18-13}
\end{equation*}
$$

where $I_{y}(x)$ is the quadratic moment of the cross-section to the $y$-axis, which is dependent on the coordinate $x$ thanks to the variable height $h(x)^{1}$. The cross-section in the elastic-plastic state is divided into the elastic core and the plastic surface, which is described by the parameter $z_{\mathrm{p}}$ according to the Figure 2.

[^0]
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$$
\begin{equation*}
z_{\mathrm{p}}=\frac{f_{\mathrm{y}}}{\kappa_{\mathrm{p}}(x) E} \tag{18-14}
\end{equation*}
$$

The elastic-plastic moment $M_{\mathrm{ep}}$ of the cross-section in the elastic-plastic state has to equal to the bending moment $M$.

$$
M_{\mathrm{ep}}(x)=2 \int_{0}^{z_{\mathrm{p}}}-\kappa_{\mathrm{p}}(x) E z^{2} w d z+2 \int_{z_{\mathrm{p}}}^{h(x) / 2}-f_{\mathrm{y}} z w \mathrm{~d} z=-\frac{q(L-x)^{2}}{2}
$$

The curvature $\kappa_{\mathrm{p}}$ in the elastic-plastic zone ( $x<x_{\mathrm{p}}$ ) results from this equality.

$$
\begin{equation*}
\kappa_{\mathrm{p}}=\frac{2 f_{\mathrm{y}}}{E \sqrt{3}} \frac{1}{\sqrt{h(x)^{2}-\frac{2 q(L-x)^{2}}{w f_{\mathrm{y}}}}} \tag{18-16}
\end{equation*}
$$

The total deflection of the structure $u_{z, \text { max }}$ is defined as a superposition of the elastic-plastic and the elastic contribution using the Mohr's integral

$$
u_{z, \max }=\int_{0}^{x_{\mathrm{p}}} \kappa_{\mathrm{p}}(L-x) \mathrm{d} x+\int_{x_{\mathrm{p}}}^{L} \kappa_{\mathrm{e}}(L-x) \mathrm{d} x=27.908+58.091=85.999 \mathrm{~mm} \quad(18-17)
$$

## RFEM 5 Settings

- Modeled in RFEM 5.16.01
- The element size is $I_{\mathrm{FE}}=0.020 \mathrm{~m}$ for files $0018.01-0018.03$ and $I_{\mathrm{FE}}=0.005 \mathrm{~m}$ for files 0018.04 and 0018.05
- Geometrically linear analysis is considered
- The number of increments is 10
- Shear stiffness of the members is neglected


## Results

| Structure File | Entity | Material model | Description |
| :---: | :---: | :---: | :---: |
| 0018.01 | Member | Isotropic Plastic 1D | - |
| 0018.02 | Plate | Isotropic Nonlinear <br> Elastic 2D | - |
| 0018.03 | Plate | Isotropic Plastic 2D/3D | - |
| 0018.04 | Plate | Isotropic Nonlinear <br> Elastic 2D | Variable Thickness |
| 0018.05 | Plate | Isotropic Plastic 2D/3D | Variable Thickness |
| 0018.06 | Member | Nonlinear Elastic 1D | - |

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| Model | Analytical Solution | RFEM 5 |  |
| :---: | :---: | :---: | :---: |
|  | $\begin{aligned} & u_{z, \max } \\ & {[\mathrm{~mm}]} \end{aligned}$ | $\begin{aligned} & u_{z, \max } \\ & {[\mathrm{~mm}]} \end{aligned}$ | Ratio [-] |
| Isotropic Plastic 1D | 85.999 | 86.215 | 1.003 |
| Isotropic Nonlinear Elastic 2D, Plate |  | 86.566 | 1.007 |
| Isotropic Plastic 2D/3D, Plate |  | 84.142 | 0.978 |
| Isotropic Nonlinear Elastic 2D, Plate, Variable Thickness |  | 83.728 | 0.974 |
| Isotropic Plastic 2D/3D, Plate, Variable Thickness |  | 83.088 | 0.966 |
| Isotropic Nonlinear Elastic 1D |  | 86.215 | 1.003 |

## References

[1] LUBLINER, J. Plasticity theory. Berkeley: University of California, 1990.


[^0]:    ${ }^{1} I_{y}=\frac{1}{12} w h^{3}(x)$

