## Category: Isotropic Linear Elasticity, Geometrically Linear Analysis, Member, Plate,

 Solid
## Verification Example: 0086 - Curved Beam with Out-of-Plane Loading

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## Description

A quarter-circle beam with rectangular cross-section $w \times h$ is loaded by means of an out-of-plane force $F$ according to Figure 1. While neglecting self-weight, determine the total deflection $u_{z}$ of the curved beam.

| Material | Modulus of <br> Elasticity | $E$ | 210000.0 | MPa |
| :--- | :--- | :--- | :--- | :---: |
|  | Poisson's Ratio | $\nu$ | 0.296 | - |
| Geometry | Radius | $r$ | 1.000 | m |
|  | Cross-section <br> Width | $w$ | 25.000 | mm |
|  | Cross-section <br> Height | $h$ | 50.000 | mm |
| Load | Force | $F$ | 1.000 | kN |



Figure 1: Problem Sketch

## Analytical Solution

The curved beam is loaded by a bending moment $M_{b}$, torsional moment $M_{t}$ and by a transverse force $T$. Considering the scheme in Figure 2, these loads at an arbitrary section are equal to

$$
\begin{align*}
M_{b} & =F a=F r \sin \varphi,  \tag{86-1}\\
M_{t} & =F b=F r(1-\cos \varphi),  \tag{86-2}\\
T & =F . \tag{86-3}
\end{align*}
$$

The deflection of the structure can be then determined according to Castigliano's second theorem

$$
\begin{equation*}
u_{z}=\frac{\mathrm{d} U}{\mathrm{~d} F}=\frac{\mathrm{d}\left(U_{b}+U_{t}+U_{s}\right)}{\mathrm{d} F} \tag{86-4}
\end{equation*}
$$

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where the total strain energy $U$ is composed of the bending strain energy $U_{b}$, torsional strain energy $U_{t}$ and shear strain energy $U_{s}$.


Figure 2: Scheme
The strain energy is calculated integrating along the length of the curved beam $L$. Considering polar coordinates, the infinitesimal length of the arc is defined as $\mathrm{d} s=r \mathrm{~d} \varphi$.

$$
\begin{align*}
& U_{b}=\int_{L} \frac{M_{b}^{2}(s)}{2 E I_{y}} \mathrm{~d} s=\int_{0}^{\pi / 2} \frac{M_{b}^{2}(\varphi)}{2 E I_{y}} r \mathrm{~d} \varphi  \tag{86-5}\\
& U_{t}=\int_{L} \frac{M_{t}^{2}(s)}{2 G J} \mathrm{~d} s=\int_{0}^{\pi / 2} \frac{M_{t}^{2}(\varphi)}{2 G J} r \mathrm{~d} \varphi  \tag{86-6}\\
& U_{s}=\frac{6}{5} \int_{L} \frac{T^{2}}{2 G A} \mathrm{~d} s=\frac{6}{5} \int_{0}^{\pi / 2} \frac{T^{2}}{2 G A} r \mathrm{~d} \varphi \tag{86-7}
\end{align*}
$$

The second moment of the area $I_{y}$ for the rectangular cross-section is defined as $I_{y}=\frac{1}{12} w h^{3}$, the torsional constant $J$ is defined as $J=0.229 h w^{3}$ because of the particular $h / w$ ratio according to [1] and the cross-section area is equal to $A=w h$. The total deflection of the curved beam $u_{z}$ is then equal to

$$
\begin{equation*}
u_{z}=\frac{3 \pi F r^{3}}{E w h^{3}}+\frac{1.555 F r^{3}}{G h w^{3}}+\frac{3 \pi F r}{5 G h w} \approx 38.960 \mathrm{~mm} \tag{86-8}
\end{equation*}
$$

## RFEM 5 Settings

- Modeled in RFEM 5.12.02
- Element size is $I_{\mathrm{FE}}=0.010 \mathrm{~m}$
- The number of increments is 10
- Isotropic linear elastic material is used
- Mindlin plate bending theory is used

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## Results

| Structure File | Entity | Orientation |
| :---: | :---: | :---: |
| 0086.01 | Member | - |
| 0086.02 | Plate | Horizontal |
| 0086.03 | Plate | Vertical |
| 0086.04 | Solid | - |


| Entity | Theory | RFEM 5 |  |
| :---: | :---: | :---: | :---: |
|  | $\begin{gathered} u_{z} \\ {[\mathrm{~mm}]} \end{gathered}$ | $\begin{gathered} u_{z} \\ {[\mathrm{~mm}]} \end{gathered}$ | Ratio <br> [-] |
| Member | 38.960 | 38.974 | 1.000 |
| Plate, horizontal |  | 38.642 | 0.992 |
| Plate, vertical |  | 38.117 | 0.978 |
| Solid |  | 38.398 | 0.986 |

## References

[1] https://www.colorado.edu/engineering/CAS/courses.d/Structures.d/, Introduction to aerospace structures

