

Structural Analysis & Design Software





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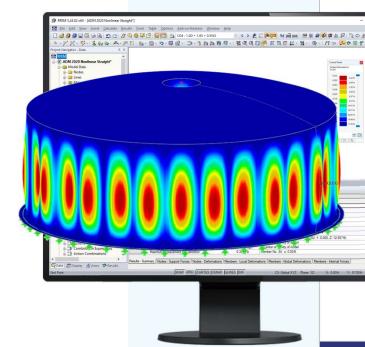


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Webinar

Stability Analysis in RFEM 6



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Questions During the Presentation



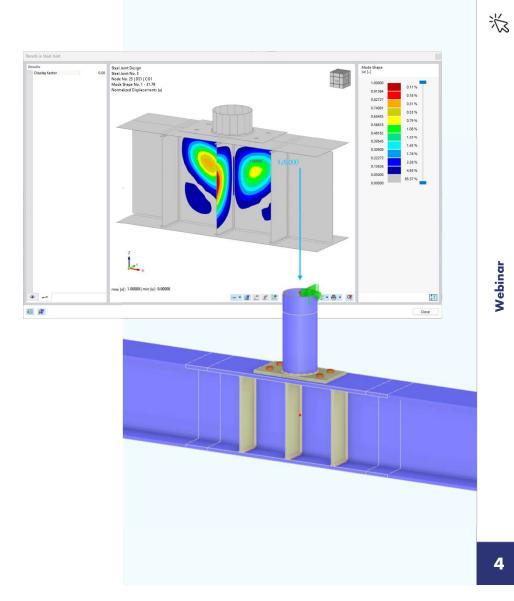






Content

- **Structure Stability Add-on advantages** 01
- Member and plate verification examples
- 3D beam model stability analysis
- 3D surface model stability analysis
- Torsional Warping (7 DOF) Add-on integration
- **Steel Joints Add-on integration**
- Instability troubleshooting



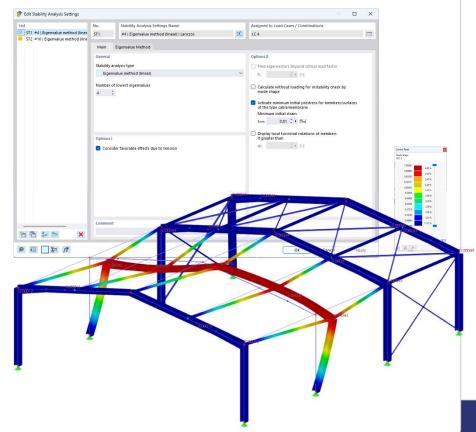


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Structure Stability Add-on Benefits

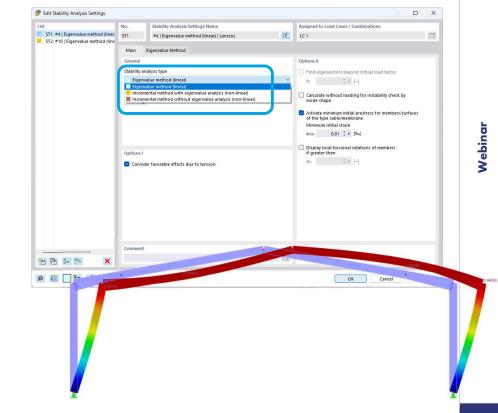
- Eigenvalue analysis to determine critical load factors and eigenvectors (buckling modes)
- Evaluate the stability of the structural system and regions prone to buckling
- Ideal for slender beams and thin-walled shells.
- Nonlinear stability analysis available using load increments
- Modeling and instability troubleshooting with loadindependent eigenvalue analysis
- Graphical rendering of stability modes
- Basis for applying structure imperfections, importing effective length factors, etc.





Structure Stability Add-on Analysis Options

- Eigenvalue method (linear)
 - Buckling modes determined through linear method
 - Nonlinearities are converted to linear elements with constant stiffnesses
- Incremental method with eigenvalue analysis (non-linear)
 - Consider all geometric and material nonlinearities
 - Failure criteria and nonlinear effects included during incremental loading increase
 - Linear stability analysis on the last stable state
- Incremental method without eigenvalue analysis (non-linear)
 - Nonlinear analysis w/o eigenvalue analysis carried out





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Euler Column Example

W 8x35

 $I_y = 127 \text{ in}^4$

 $I_z = 42.6 \text{ in}^4$

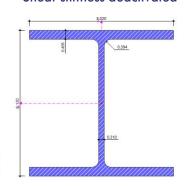
E = 29,000 ksi

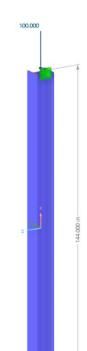
L = 144 in

Support $1 = \mathbf{\nabla} \mathbf{u} \mathbf{X}, \mathbf{u} \mathbf{Y}, \mathbf{u} \mathbf{Z}, \mathbf{\Phi} \mathbf{Z}$

Support $2 = \mathbf{\nabla} \mathbf{u} \mathbf{X}, \mathbf{u} \mathbf{Y}$

*Shear stiffness deactivated





Critical Load	Analytical	RFEM 6	Ratio
N _{cr,y} (strong)	1752.98 kip	1752.70 kip	1.000
N _{cr,z} (weak)	588.00 kip	588.00 kip	1.000

$$N_{cr,y} = \frac{\pi^2 EI}{L^2}$$

$$N_{cr,y} = \frac{\pi^2 (29000 \, ksi) (127 \, in^4)}{(144 \, in)^2}$$

$$N_{cr,y} = 1752.98 \, kip$$

$$N_{cr,z} = \frac{\pi^2 EI}{L^2}$$

$$N_{cr,z} = \frac{\pi^2 (29000 \text{ ksi}) (42.6 \text{ in}^4)}{(144 \text{ in})^2}$$

$$N_{cr,z} = 588.00 \text{ kip}$$



Euler Shell Example

E = 29,000 ksi

v = 0.3

b = 10 in

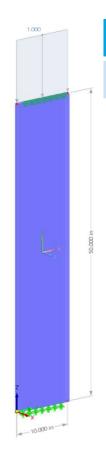
L = 50 in

t = 1 in

 $I_x = 0.833 \text{ in}^4$

Support $1 = \square uX, uY, uZ$

Support 2 = ☑ uX, uY



Critical Load	Analytical	RFEM 6	Ratio
N _{cr} /b	9.54 kip/in	9.56 kip/in	0.997

$$N_{cr} = \frac{\pi^2 EI}{L^2}$$

$$N_{cr} = \frac{\pi^2 (29000 \text{ ksi}) (0.833 \text{ in}^4)}{(50 \text{ in})^2}$$

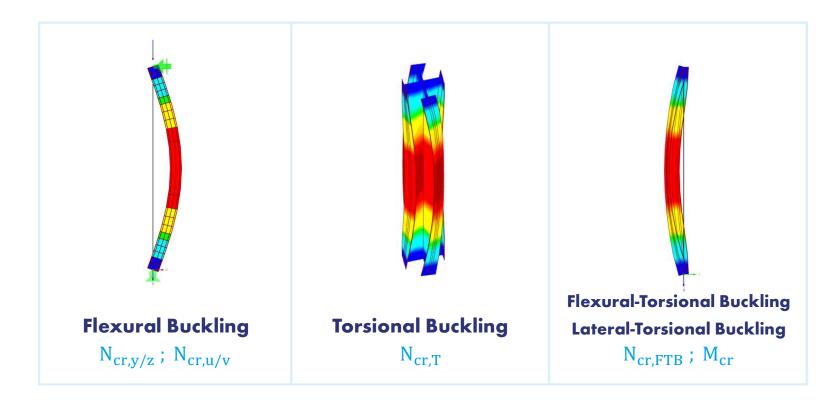
$$N_{cr} = 95.41 \text{ kip}$$

$$\frac{N_{cr}}{b} = \frac{95.41 \text{ kip}}{10 \text{ in}}$$

$$\frac{N_{cr}}{b} = 9.54 \text{ kip/in}$$



Stability Failure Modes

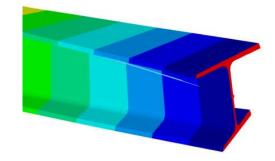




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Torsional Warping (7 DOF) Add-on Integration

- Consideration to cross-sectional warping with an additional DOF for member calculations
- Integration with Structural Stability Add-on to determine additional member critical load factors and mode shapes
 - 6 DOF can only consider flexural buckling failure mode shapes; load application at shear center
 - 7 DOF can additionally capture torsional buckling, flexural-torsional buckling, and lateral torsional buckling; load application at center of gravity
- Additional options available such as "Member Transverse Stiffeners"
- Future development to consider 7 DOF in Steel Design Addon acc. to AISC and CSA \$16 standards





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