#### Program: RFEM 5

**Category:** Geometrically Linear Analysis, Isotropic Nonlinear Elasticity, Isotropic Plasticity, Isotropic Masonry, Member, Plate

Verification Example: 0021 – Plastic Bending with Zero Tensile Strength

# 0021 – Plastic Bending with Zero Tensile Strength

# Description

A cantilever is fully fixed on the left end (x = 0) and subjected to a transverse force F and an axial force  $F_a$  on the right end according to the **Figure 1**. The tensile strength is zero and the behaviour in the compression remains elastic. The problem is described by the following set of parameters.

Material	Elastic-Plastic	Modulus of Elasticity	Ε	210000.000	MPa
		Poisson's Ratio	ν	0.000	—
		Shear Modulus	G	105000.000	MPa
		Tensile Plastic Strength	f <sub>t</sub>	0.000	MPa
Geometry	Cantilever	Length	L	2.000	m
		Width	W	0.005	m
		Thickness	t	0.005	m
Load		Transverse Force	F	4.000	Ν
		Axial Force	Fa	5000.000	N

Small deformations are considered and the self-weight is neglected in this example. Determine the maximum deflection  $u_{z,max}$ .



Figure 1: Problem sketch

# **Analytical Solution**

The bending moment *M* for the cantilever under transverse force *F* is defined as

$$M = -F(L-x) \tag{21-1}$$



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#### Linear Analysis

Considering linear analysis (only elasticity) the maximum deflection of the structure can be calculated as follows:

$$u_{z,\text{max}} = \frac{FL^3}{3El_v} \approx 0.975 \text{ m} \tag{21-2}$$

Nonlinear Analysis

The transverse force F together with the axial force  $F_a$  causes the elastic-plastic of the cantilever according to the **Figure 1**. The elastic-plastic zone length is described by the parameter  $x_p$ . The stress  $\sigma_x$  is composed of the bending stress  $\sigma_b$  and the compressive stress  $\sigma_c$  according to the **Figure 2** and it is defined according to the following formula

$$\sigma_x(x,z) = -\kappa(x)E(z-z_0(x)) \tag{21-3}$$

where  $\kappa(x)$  is the curvature defined as  $\kappa(x) = d^2 u_z/dx^2$  [1] and the parameter  $z_0(x)$  is defined so that  $\sigma_x(x, z_0) = 0$ , see **Figure 2**. The first yield occurs, when the bending stress on the top surface on the fixed end reaches the value of the compressive stress.

$$\frac{M(0)}{l_v}\frac{t}{2} = \frac{F_a}{A} \tag{21-4}$$

where  $I_y$  is the quadratic moment of the cross-section to the *y*-axis<sup>1</sup> and *A* is the area of the cross-section<sup>2</sup>. The transverse force results  $F = 2.08\overline{3}$  N. Thus, the the cantilever under transverse force F = 4.000 N is in elastic-plastic state.



Figure 2: Stress distribution

$$I_y = \frac{1}{12}wt^3 = 52.08\bar{3} \text{ mm}^4$$

$$A^{2} \dot{A} = wt = 25.000 \text{ mm}^{2}$$

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In the elastic-plastic zone ( $x < x_p$ ) the equilibrium between bending moments and axial forces has to be satisfied.

$$M_{\rm ep} = \int_{z_0}^{t/2} -\kappa_{\rm p} E(z - z_0) z w \, \mathrm{d}z = M \tag{21-5}$$

$$N_{\rm ep} = \int_{z_0}^{t/2} -\kappa_{\rm p} E(z - z_0) w \, \mathrm{d}z = -F_{\rm a}$$
(21-6)

Solving equations (21 – 5) and (21 – 6) the curvature in the elastic-plastic zone  $\kappa_p$  and the parameter  $z_0$  results as follows.

$$\kappa_{\rm p}(x) = \frac{8F_{\rm a}^3}{9Ew(F_{\rm a}t + 2M)^2}$$
(21 - 7)

$$z_0(x) = \frac{t}{2} - \frac{3}{2} \frac{F_a t + 2M}{F_a}$$
(21-8)

The elastic-plastic zone length  $x_p$  can be obtained from the equation (21 – 8) under the condition  $z_0(x_p) = -t/2$ .

$$x_{\rm p} = L - \frac{tF_{\rm a}}{6F} \approx 958.333 \,\,{\rm mm}$$
 (21 – 9)

The curvature  $\kappa_{\rm e}$  in the elastic zone ( $x > x_{\rm p}$ ) is described by the Bernoulli-Euler formula

$$\kappa_{\rm e} = -\frac{M}{EI_{\rm v}} \tag{21-10}$$

The maximum deflection  $u_{z,max}$  can be finally calculated according to the following formula

$$u_{z,\max} = \int_{0}^{x_{p}} \kappa_{p}(L-x) dx + \int_{x_{p}}^{L} \kappa_{e}(L-x) dx \approx 1.232 \text{ m}$$
(21 - 11)

### **RFEM 5 Settings**

- Modeled in RFEM 5.16.01
- The element size is  $I_{\rm FE} = 0.020$  m
- Geometrically linear analysis is considered



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- The number of increments is 10
- Shear stiffness of the members is neglected

# Results

Structure File	Entity	Material model	Hypothesis
0021.01	Member	lsotropic Nonlinear Elastic 1D	-
0021.02	Plate	Isotropic Masonry 2D	-
0021.03	Plate	Nonlinear Elastic 2D/3D	Mohr-Coulomb
0021.04	Plate	Nonlinear Elastic 2D/3D	Drucker-Prager
0021.05	Plate	Isotropic Plastic 2D/3D	Mohr-Coulomb
0021.06	Plate	lsotropic Plastic 2D/3D	Drucker-Prager

Model	Analytical Solution	RFEM 5	
	u <sub>z,max</sub> [m]	u <sub>z,max</sub> [m]	Ratio [-]
Isotropic Nonlinear Elastic 1D		1.230	0.998
Isotropic Masonry 2D		1.237	1.004
Nonlinear Elastic 2D/3D, Mohr-Coulomb		1.237	1.004
Nonlinear Elastic 2D/3D, Drucker-Prager	1.232	1.237	1.004
lsotropic Plastic 2D/3D, Mohr-Coulomb		1.237	1.004
lsotropic Plastic 2D/3D, Drucker-Prager		1.236	1.003

# References

[1] LUBLINER, J. *Plasticity theory*. Berkeley: University of California, 1990.

